,	
γ	
7	
6111	

REPORT DO	CUMENTATIO	ON PAGE	i	Form Approved
ublic reporting burden for this collection of information is a	atimated to assess 4 b		reviewing instructions.	OMB No. 0704-0188
actuding suggestions for reducing this burden to Deserte	A = 1 D = 1	the seminante regulating this be	arcieri estimate of any (DIRECT ASDECT OF this collection of information
flighway, Suite 1204, Arlington, VA 22202-4302. Responding to the formation of it does not display a currently valid	ents should be aware that notwi	thstanding any other provision o	f law, no person shall b	be subject to any penalty for failing to comply with
I. REPORT DATE (DD-MM-YYYY)	2. REPORT TYPE	LEGONOT RETORN TOUR FO	HAM TO THE ABOVE	3. DATES COVERED (From - To)
	Technical Papers			: DATES COVERED (FIONT - 10)
I. TITLE AND SUBTITLE				5a. CONTRACT NUMBER
			L	;
				5b. GRANT NUMBER
			_	
				5c. PROGRAM ELEMENT NUMBE
. AUTHOR(S)				
			,	5d. PROJECT NUMBER
		•	L.	2362
	,		[]	5e. TASK NUMBER
			<u> </u>	MIG 2
			'	of. WORK UNIT NUMBER
PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)			DEDEODMING ODG ANIZATION
		•		B. PERFORMING ORGANIZATION REPORT
ir Force Research Laboratory (AFMC))		1	
FRL/PRS		•		1
Pollux Drive				1
dwards AFB CA 93524-7048	2		ſ	1
SPONSORING / MONITORING AGENCY	NAME(S) AND ADDRE	SS(ES)		10. SPONSOR/MONITOR'S
	•	• •	· · · / / /	ACRONYM(S)
ir Force Decemb I should be to the			-	
ir Force Research Laboratory (AFMC)	· .		1	1
FRL/PRS] 1	1. SPONSOR/MONITOR'S
Pollux Drive dwards AFB CA 93524-7048			1	NUMBER(S)
2. DISTRIBUTION / AVAILABILITY STATES				
pproved for public release; distribution	unlimited.			
. SUPPLEMENTARY NOTES				1
				į
ABSTRACT			* * * * * * * * * * * * * * * * * * * *	:
				i ,
				1
		•		
	•			•
SUBJECT TERMS				
- CONTRACT IN THE CONTRACT IN				
SECURITY CLASSIFICATION OF:		17. LIMITATION	18. NUMBER	100 NAME OF DECEMBER
		OF ABSTRACT	OF PAGES	19a. NAME OF RESPONSIBLE PERSON
				Leilani Richardson
EPORT b. ABSTRACT	c. THIS PAGE	1		19b. TELEPHONE NUMBER
alander, y		(A)	1	(include area code)
classified Unclassified	Unclassified			(661) 275-5015
				Standard Form 298 (Rev. 8-98)
36 Sepa	ate den	s are e	1 dese	Prescribed by ANSI Std. 239.18

TP-FY99-0134

6

/Sprendsheet

MEMORANDUM FOR PRS (Contractor/In-House Publication)

FROM: PROI (TI) (STINFO)

16 June 1999

SUBJECT: Authorization for Release of Technical Information, Control Number: AFRL-PR-ED-TP-FY99-0134 C.T. Liu, "Influence of Near Tip Damage on the Initiation Fracture Toughness of a Particulate Composite"

1999 ASME Summer Conference

Presentation

(Public Release)

Influence of Near Tip Damage on the Initiation Fracture Toughness of a Particulate Composite

C. T. Liu

And

T. Miller

OL-AC AFRL/PRSM

<----

10 E. Saturn Blvd. Edwards AFB CA 93524-7680

20021119 127

Objectives

Investigate the Effects of Crack Tip damage, Specimen Thickness, and Initial Crack length on the Initiation Fracture Toughness of a Particulate Composite Material.

Specimen Thickness: 0.2 in, 0.5 in, 1.0 in, and 1.5 in.

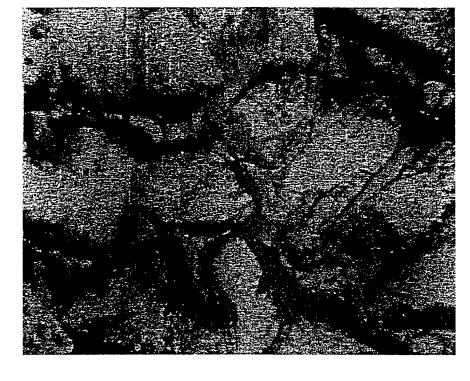
Initial Crack Length: 0.1 in; 0.2 in; 0.3 in; and 0.4 in.



Local Dewetting About Filler Particles in Propellant

Direction of Strain

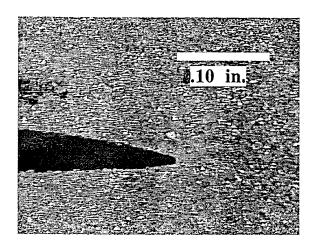




30% Strain

Ao Ao 3 in.

Specimen Geometry



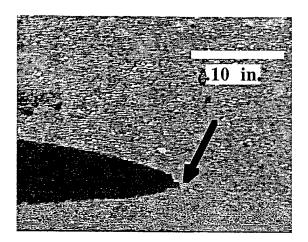


Figure xx - Crack initiation, 1.0" thickness, .30" initial crack length.

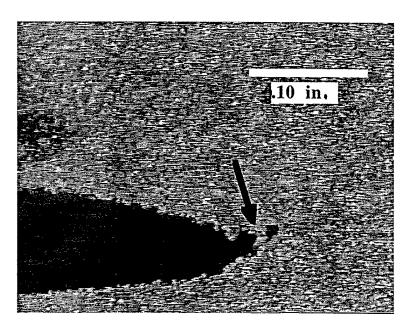


Figure xx - Ligament formation, 1.50%, thickness.

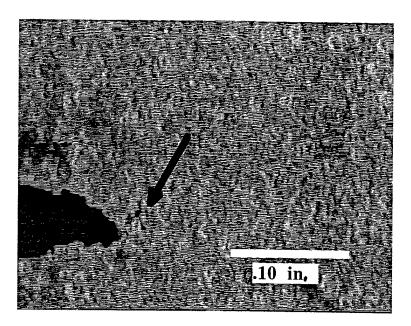


Figure xx - Damaged region ahead of crack tip, 1.5% thickness.

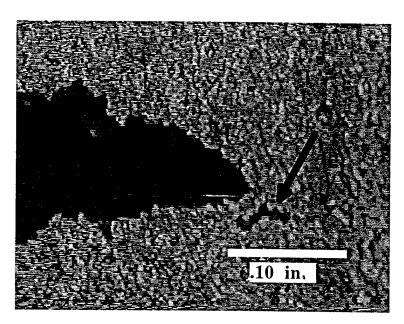


Figure xx - Damaged region ahead of crack tip 12: thickness.

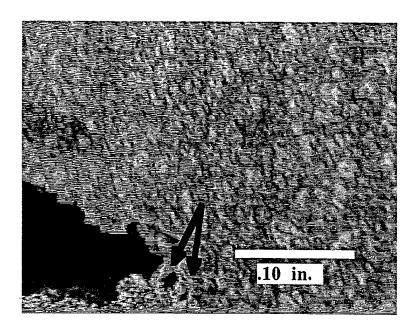


Figure xx - Double ligament formation , 50% thickness.

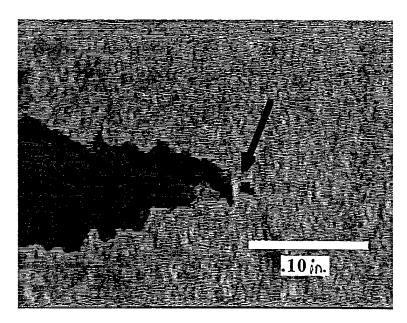
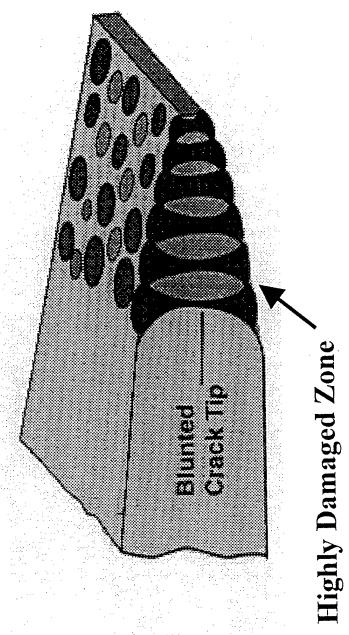
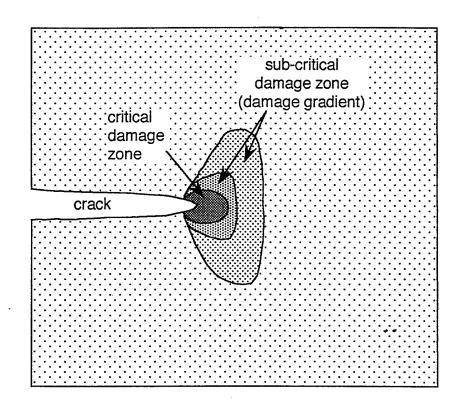


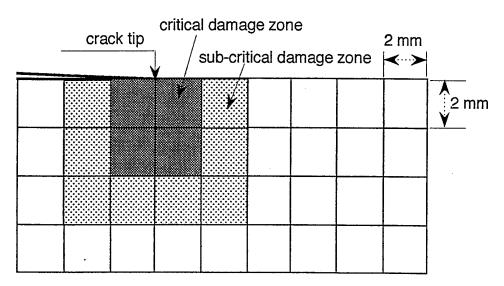
Figure xx - Ligament formation, 220; thickness.



Crack Tip Damage Model



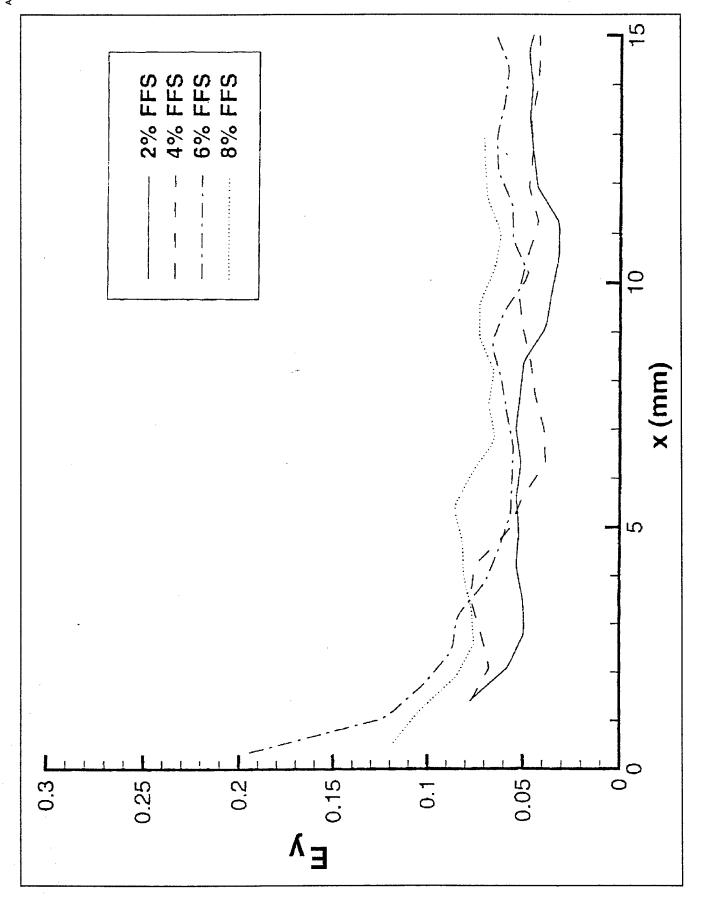




FEM mesh at crack tip

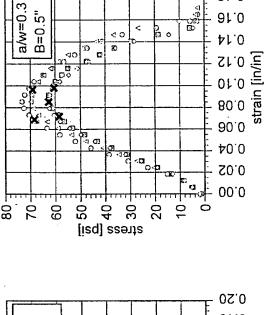
Table 1. Summary of crack-damage interaction analysis

1		Die 1. Dullillai	analysis				
Case	Damaga Elomant	Damage	Poisson's	Inside	Middle	Outside	_
Case	Damage Element	Element	Ratio	Layer	Layer	Layer	
		Modulus		K,	K	K	-
41		MPa (psi)		MPa-cm ^{0.5}	MPa-cm ^{0.5}	MPa-cm ^{0.5}	1
thin*	none	0.414 (60)	0.4999	1.871		= 5	\dashv
	none		0.4999	1.931	1.903	1.802	4
2	325, 297	0.414 (60)	0.4999	0.422	2.246		\dashv
3	325, 297	0.414 (60)	0.4999	0.535	0.440	1.871	4
	326, 298			0.000	0.440	2.208	
4	325, 297	0.414 (60)	0.4999	0.573	0.524	0.155	1
	326, 299	` ′		0.575	0.524	0.455	
	327, 299						
5	325, 297, 322, 294	0.414 (60)	0.4999	0.392	2 205	·	
6	325, 297, 322, 294	0.414 (60)	0.4999		2.285	1.906	
-	326, 298, 323, 295		0.4999	0.522	0.497	0.432	
	327, 299, 324, 296				ř		
7	325, 297, 322, 294	0.414 (60)	0.4999	0.546	0.544		
-	326, 298, 323, 295	(66)	0.4555	0.546	0.514	0.442	
	327, 299, 324, 296						ĺ
	353, 350, 347, 319	0.828 (120)	0.4999				
	291, 263, 266, 269	(:20)	0.4000				
	354, 351, 349, 320	İ					
	292, 264, 267, 270		1				
	355, 352, 349, 321		{	1	į		
	293, 265, 268, 271					-	



Fracture Specimens a/w = 0.3

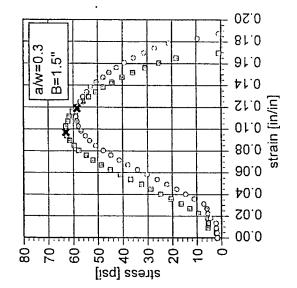


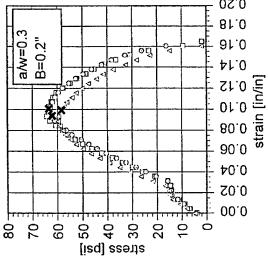


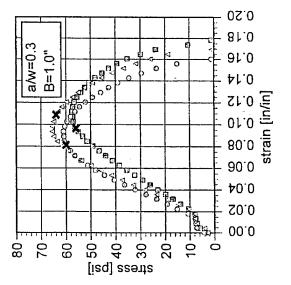
0.20

81.0

91.0







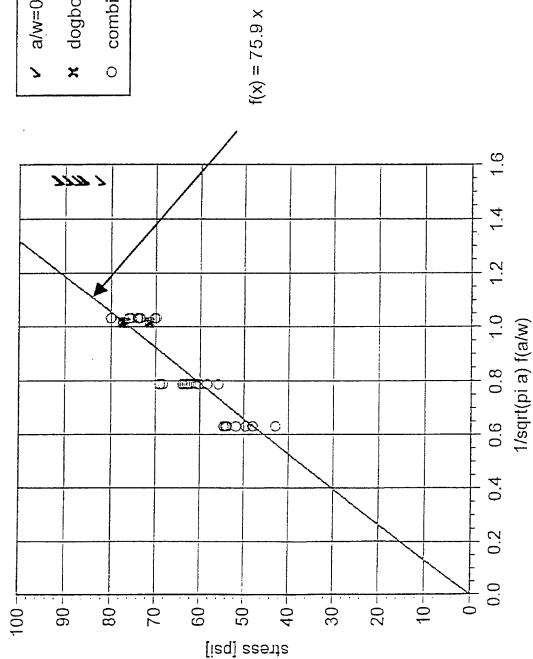


Regressive Calculation of K

Slope of Curve Gives Value for K



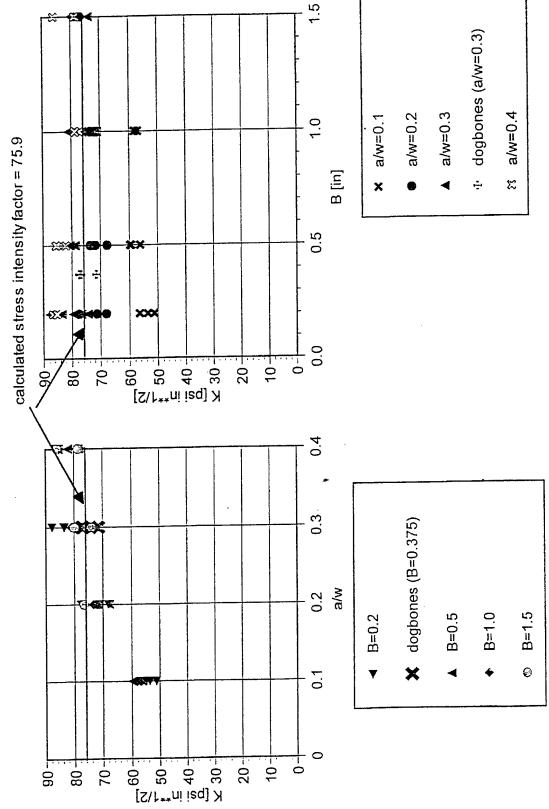
- dogbones (a/w=0.3)
- combined data



 $R^{4} = 0.88$

Variations o

Variations of Fracture Toughness at Crack Initiation with a/w, Thickness (Ambient Pressure)



Sp

Specimen* (Ambient Pressure, Strain Rate = 8 min-1) Table I Summary of K_{li} Value for Sheet

_		ŗ			Psi√in.
K _{li} Average psi √ n	55.63	70.43	77.15	82.94).4 = 76.84 Psi√in.
15	52.46	66.10	74.20	83.51	(K _{li}) 0.2, 0.3, 0.4 Average
1.0	57.75	72.15	76.08	78.14	<u> </u>
0.5	58.51	71.10	77.92	84.34	-
Specimen 0.2 Thickness (in)	53.81	72.38	80.41	85.76	<u>-</u>
Initial Space Grack ength Ao	(in) 0.1	0.2	0.3	0.4	- -

	3 in.	
1 in,	Ao	

Conclusions

- 1. Local Damage at the Crack Tip Minimized the Transverse Constraint.
- 2. The Initiation Mode I Fracture Toughness $K_{\rm IC\ I}$ is Insensitive to the Specimen Thickness.
- 3. Linear Fracture Mechanics can be Used to Determine $K_{\rm IC\,I}$ for Initial Crack Length Equal to or Greater than 0.2 in.

PLAIN

4. There is no Plane Strain Fracture Toughness of this Particulate Composite Material.